

Errata

Fuselage Boundary-Layer Effects on Sound Propagation and Scattering

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DUE to an error during the production of this paper, Eq. (3) was published incorrectly. The correct Eq. (3) is

$$p\omega = \sum_{n=-\infty}^{\infty} e^{in\theta} \int_{-\infty}^{\infty} F(r;n,k_x,\omega) e^{ik_x x} dk_x$$

Navier-Stokes Solutions for Vortical Flows over a Tangent-Ogive Cylinder

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THE following revised figures should replace those published with this article:

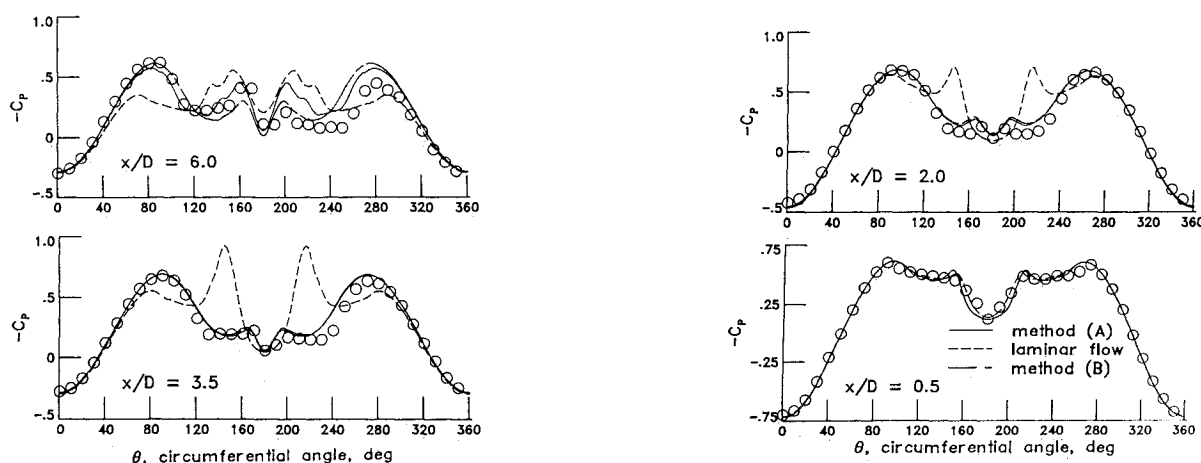


Fig. 6 Experimental (symbols) and computational (lines) circumferential surface pressure distributions: $\alpha = 20$ deg, $Re_D = 0.2$ million, $40 \times 73 \times 65$ grid points.

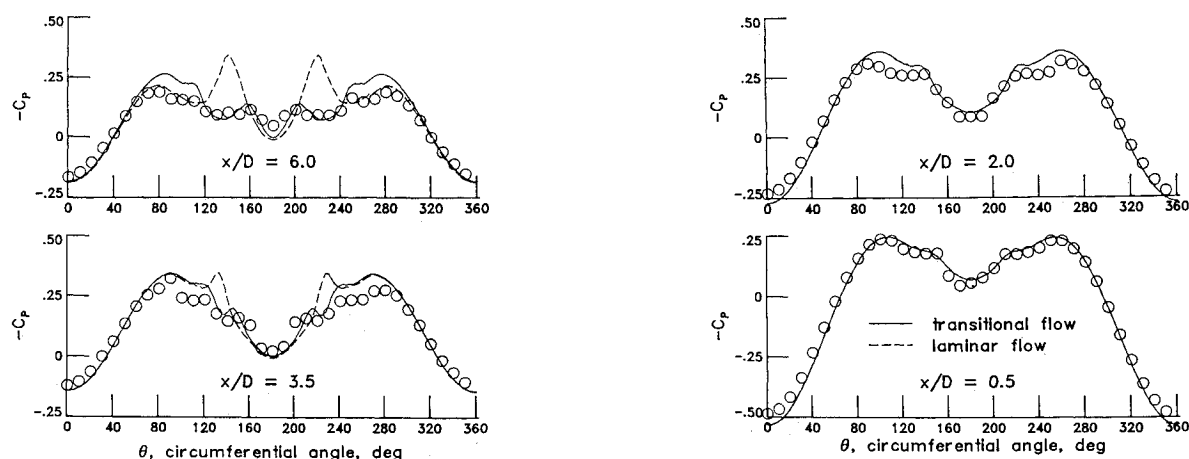


Fig. 8 Measured (symbols) and computed (lines) circumferential surface pressure distributions: $\alpha = 30$ deg, $Re_D = 0.8$ million, $40 \times 73 \times 65$ grid points.